

Dimensional Analysis

102  
510 Buckingham's  $\Pi$  theorem

$$\phi(Q_1, Q_2, Q_3, \dots, Q_m) = 0$$

Where  $Q_1, Q_2, Q_3, \dots, Q_n$  are dimensional physical quantities then there can be  $(n-m)$  dimensionless  $\Pi$  quantities that describe the same phenomenon as

$$f(\Pi_1, \Pi_2, \Pi_3, \dots, \Pi_{n-m}) = 0$$

Proof:-

The no. of dimensionless required to sepecify completely the relationship is the number of variable  $n$ , the number of dimensions  $m$  involved in the variables.

Consider  $\phi(Q_1, Q_2, Q_3, Q_4, Q_5)$

$$\phi(Q_1, Q_2, Q_3, Q_4, Q_5) = 0 \quad \text{--- (1)}$$

(1) Expressed form of a Power series:

$$\begin{aligned} \phi(Q_1, Q_2, Q_3, Q_4, Q_5) = & A_1(Q_1^\alpha, Q_2^\beta, Q_3^\gamma, Q_4^\delta, Q_5^\epsilon) \\ & + A_2(Q_1^\alpha, Q_2^\beta, Q_3^\gamma, Q_4^\delta, Q_5^\epsilon) \\ & \dots + A_k(Q_1^\alpha, Q_2^\beta, Q_3^\gamma, Q_4^\delta, Q_5^\epsilon) \end{aligned}$$

Where  $A_k$ 's are dimensionless coefficient

And the exponents  $\alpha, \beta, \gamma, \delta, \epsilon$  are integer

Let All term contain the same dimensions

$$(Q_1^\alpha Q_2^\beta Q_3^\gamma Q_4^\delta Q_5^\epsilon)^k = M^{ka} L^{kb} T^{kc}$$

For equal dimensions

$$M^a L^b T^c = M^{ka} L^{kb} T^{kc}$$

$$\Rightarrow a = b = c = 0, \text{ for } k \neq 0$$

It follows that every term (2) must be dimensionless.

$$(Q_1^\alpha Q_2^\beta Q_3^\gamma Q_4^\delta Q_5^\epsilon) = M^a L^b T^c \text{ has zero dimension.} \quad \text{--- (3)}$$

$$Q_1 = M^{a_1} L^{b_1} T^{c_1}$$

$$Q_2 = M^{a_2} L^{b_2} T^{c_2}$$

$$Q_5 = M^{a_5} L^{b_5} T^{c_5} \quad \text{--- (4)}$$

From (3) And (4) we have

$$(M^{a_1} L^{b_1} T^{c_1})^\alpha (M^{a_2} L^{b_2} T^{c_2})^\beta (M^{a_3} L^{b_3} T^{c_3})^\gamma (M^{a_4} L^{b_4} T^{c_4})^\delta (M^{a_5} L^{b_5} T^{c_5})^\epsilon = M^0 L^0 T^0$$

Equating the Power of M, L, T we have

$$a_1 \alpha + a_2 \beta + a_3 \gamma + a_4 \delta + a_5 \epsilon = 0 \quad \text{--- (5)}$$

$$b_1 \alpha + b_2 \beta + b_3 \gamma + b_4 \delta + b_5 \epsilon = 0 \quad \text{--- (6)}$$

$$c_1 \alpha + c_2 \beta + c_3 \gamma + c_4 \delta + c_5 \epsilon = 0 \quad \text{--- (7)}$$

Since there are only three equations, we may solve three exponents  $\alpha, \beta, \gamma$ , in terms of the others.

$$a_1 \alpha + a_2 \beta + a_3 \gamma = -a_4 \delta - a_5 \epsilon \quad \text{--- (8)}$$

$$b_1 \alpha + b_2 \beta + b_3 \gamma = -b_4 \delta - b_5 \epsilon \quad \text{--- (9)}$$

$$c_1 \alpha + c_2 \beta + c_3 \gamma = -c_4 \delta - c_5 \epsilon \quad \text{--- (10)}$$

Now  $\alpha, \beta, \gamma$  can be determinant in term of the other  $\delta, \epsilon$

Solving Equation (8) (9) (10) we have

$$\gamma = p_3 \delta + q_3 \epsilon$$

Similarly other two exponents written as

$$\alpha = p_1 \delta + q_1 \epsilon$$

And

$$\beta = p_2 \delta + q_2 \epsilon$$

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10.3 Similitude:->

Two flows are said to be mechanically similar. they are

- (i) Kinematically
- (ii) Dynamically
- (iii) Geometrically

two flows are said to be geometrically similar, if they can be made to appear photographically alike. i.e. they differ in their absolute size.

The ratios of corresponding length dimensions between the model and prototype must be same. the ratio of characteristic length  $l_1$  is a constant at corresponding parts of two  $l_2$  flow fields.

Kinematic similarity shows similarity of motion. the ratios of the velocity and acceleration of two flows must be same at the corresponding point. the magnitude and direction of velocity and acceleration at the corresponding point in the two flows, should be the same. the streamline will, therefore, appear similar two

flows. we have  $\frac{q}{q'} = \frac{q_p}{q'_p}$

And  $\frac{f}{f'} = \frac{f_p}{f'_p}$

Geometric similarity is a necessary condition but not a sufficient condition for Kinematic similarity.

Dynamic Similarity:- It is a  
consider the equation of motion for the two  
flows.

$$\rho_1 \frac{dq_1}{dt_1} = \rho_1 q_1 - \nabla_1 p_1 + \mu_1 \nabla_1^2 q_1$$

$$\rho_2 \frac{dq_2}{dt_2} = \rho_2 q_2 - \nabla_2 p_2 + \mu_2 \nabla_2^2 q_2$$

For the flows to be dynamically similar, the  
corresponding ratio

$$\frac{\rho_2 \frac{dq_2}{dt_2}}{\rho_1 \frac{dq_1}{dt_1}} = \frac{\rho_2 q_2}{\rho_1 q_1} = \frac{\nabla_2 p_2}{\nabla_1 p_1} = \frac{\mu_2 \nabla_2^2 q_2}{\mu_1 \nabla_1^2 q_1}$$

(i)                      (ii)                      (iii)                      (iv)

I Froude Number:- we know

$$\frac{\rho_2 \frac{dq_2}{dt_2}}{\rho_1 \frac{dq_1}{dt_1}} = \frac{\rho_2 q_2}{\rho_1 q_1}$$

$$\Rightarrow \frac{\rho_2 \frac{dq_2}{dt_2}}{\rho_2 q_2} = \frac{\rho_1 \frac{dq_1}{dt_1}}{\rho_1 q_1}$$

$$\Rightarrow \frac{\rho_2 \left( \frac{U_2}{L_2 U_2} \right)}{\rho_2 q_2} = \frac{\rho_1 \left( \frac{U_1}{L_1 U_1} \right)}{\rho_1 q_1}$$

$$\Rightarrow \boxed{\frac{U_2^2}{L_2 q_2} = \frac{U_1^2}{L_1 q_1} = Fr_2}$$

It is known as Froude number

Physically this represents the rate of change  
of velocity at each point depending on the  
initial boundary conditions.

For variable density flows.

$$\frac{\text{Material Velocity}}{\text{Sound Velocity}} = \frac{U_1}{a_1} = \frac{U_2}{a_2} = M$$

where  $M$  is known as Mach Number.

Physically this defines number of the compressibility of the fluid due to high speed, when the Mach number is nearly one or greater than one, the fluid will be taken as compressible.

Reynolds number:- The Reynolds number  $Re$

$$Re = \frac{\text{Inertial forces}}{\text{Viscous forces}}$$

We know

$$\Rightarrow \frac{\rho_2 (dq_2/dt_2)}{\rho_1 (dq_1/dt_1)} = \frac{\mu_2 \nabla_2^2 q_2}{\mu_1 \nabla_1^2 q_1}$$

$$\Rightarrow \frac{\rho_2 (dq_2/dt_2)}{\mu_2 \nabla_2^2 q_2} = \frac{\rho_1 (dq_1/dt_1)}{\mu_1 \nabla_1^2 q_1}$$

$$\Rightarrow \frac{\rho_2 U_2 (L_2/U_2)}{\mu_2 U_2 / L_2^2} = \frac{\rho_1 U_1 (L_1/U_1)}{\mu_1 U_1 / L_1^2}$$

$$\Rightarrow \frac{\rho_2 U_2^2 / L_2}{\mu_2 U_2 / L_2^2} = \frac{\rho_1 U_1^2 / L_1}{\mu_1 U_1 / L_1^2}$$

$$\Rightarrow \frac{\rho_2 U_2 L_2}{\mu_2} = \frac{\rho_1 U_1 L_1}{\mu_1} = Re$$

which is called Reynolds number.

$$Re = \frac{UL}{\nu}$$

✓ Reynolds no is a measure of the relative importance of dynamic forces and viscous forces.